## **Homework #3 Solutions**

Course: Classical Mechanics (Physics 603), Prof. Weinstein Spring 2021

## **Question 1**

1D particle with force  $F = -kx + ax^{-3}$  where k, a > 0. Find equilibria and the period of small oscillations.

**Answer.** a: The equilibria occur at  $F = -(\partial V / \partial x) = 0$  so that

$$-kx + ax^{-3} = 0$$

which gives

$$x = \pm^4 \sqrt{a/k}$$

(real roots only).

Checking stability:

$$\frac{\partial^2 V}{\partial x^2} = -\frac{\partial F}{\partial x} \tag{1}$$

$$= k + 3ax^{-4} \tag{2}$$

$$= k + 3a\frac{k}{a} = 4k > 0 \tag{3}$$

**b:** small oscillations

$$V(x) = V(x_0) + \frac{\partial V}{\partial x} \delta x + \frac{1}{2} \frac{\partial^2 V}{\partial x^2} (\delta x)^2 + \dots$$
(4)

$$= V_0 + \frac{1}{2}(4k)(\delta x)^2 + \dots$$
 (5)

This is a harmonic oscillator with frequency  $\omega = \sqrt{4k/m}$  for both minima.

## **Question 2**

Two masses in circular orbit around their common CM with period  $\tau$ . Their motion is suddenly stopped and they fall directly toward each other. Prove that they hit at time  $T = \tau/(4\sqrt{2})$ .

**Answer.** This reduces to one particle with relative motion described by  $\mathcal{L} = \frac{1}{2}\mu \vec{r}^2 - \frac{G\mu M}{r}$  in a circular orbit of radisu *R*. The period of a circular or elliptical orbit is

$$\tau = 2\pi a^{3/2} \sqrt{\frac{M}{k}}$$

where *a* is the semi major axis.

Let's look at this in three ways. First, as we slowly reduce the orbital speed of the mass from  $r\dot{\theta}$  to 0, the orbit becomes elliptical with *e* increasing from 0 to 1. As this happens, the aphelion distance stays the same (*R*) and the perihelion distance decreases from *R* to 0. This means that the semi major axis *a* decreases from *R* to *R*/2 as the "diameter" of the orbit decreases from 2*R* to *R*. This will decrease the period  $\tau$  by a factor of  $2\sqrt{2}$ . Since the period is the time it takes for the particle to fall from aphelion to perihelion and back, the collision time will be half the period. Thus

$$T = \frac{\tau}{4\sqrt{2}}.$$

Second, let's use more equations. The energy in a circular orbit is

$$E=T+V=\frac{1}{2}V<0.$$

When the motion is stopped, then

$$E' = T + V = 0 + V = 2E < 0$$

and the energy is doubled. Using Eq. 3.61,

$$a = -\frac{k}{2E} = R \tag{6}$$

$$a' = -\frac{k}{2E'} = \frac{1}{2}a$$
(7)

This confirms our intuition from the first discussion that the semi major axis will decrease by a factor of two.

Third, let's brute force it. The total energy when stopped is

$$E = -\frac{GM\mu}{R}$$

and energy conservation gives

$$\frac{1}{2}\mu\dot{r}^2 - \frac{GM\mu}{r} = -\frac{GM\mu}{R}$$

so that

$$\dot{r} = -\sqrt{\frac{2GM}{R}}\sqrt{\frac{R-r}{r}}$$

where the minus sign indicates that *r* is decreasing. Solving for *dt* and integrating:

$$T = -\sqrt{\frac{2GM}{R}} \int_{R}^{0} \sqrt{\frac{r}{R-r}} dr$$

Using the substitution  $r = R \sin^2 \xi$ , and reversing the integration limits:

$$\int_0^R \sqrt{\frac{r}{R-r}} dr = \int_0^{\pi/2} \sqrt{\frac{R}{R-R\sin^2 \xi}} \sin \xi R 2 \sin \xi \cos \xi d\xi$$
(8)

$$= 2R \int_0^{\pi/2} \sin^2 \xi d\xi \tag{9}$$

$$= 2R \int_0^{\pi/2} (1 - \cos(2\xi)) d\xi$$
 (10)

$$= R[\xi + \frac{1}{2}\sin(2\xi)]|_{0}^{\pi/2}$$
(11)

$$= \frac{1}{2}\pi R \tag{12}$$

which yields

$$T = \sqrt{\frac{R}{2GM}} \frac{\pi R}{2} = 2\pi \sqrt{\frac{R^3}{GM}} \frac{1}{4\sqrt{2}} = \frac{\tau}{4\sqrt{2}}$$

## **Question 3**

Find the ratio of the maximum and minimum velocities for a planet circling the Sun with eccentricity *e*.

**Answer.** We know the maximum and minimum distances are  $r_{ap} = a(1 + e)$  and  $r_{peri} = a(1 - e)$ . We also know that  $\dot{r} = 0$  at these extrema so the velocity is perpendicular to  $\vec{r}$ . This means that the angular momentum at both locations is l = mvr so that

$$mv_{ap}r_{ap} = mv_{peri}r_{peri}$$

which gives

$$\frac{v_{peri}}{v_{ap}} = \frac{r_{ap}}{r_{peri}} = \frac{1+e}{1-e}$$

and the planet is moving fastest at perihelion where *r* is minimum.

For the Earth orbiting the Sun, this amounts to

$$\frac{r_{ap}}{r_{peri}} = \frac{1+e}{1-e}|_{Earth} = \frac{1+0.0167}{1+0.0167} = 1.034$$

and a 3.4% difference in speed is quite noticeable.